General relativity tests in a dynamical model of the Solar system

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Outline

Checking for two deviations from GR in lunar-planetary numerical ephemeris:

- Equivalence principle (by Earth-Moon-Sun dynamics)
 - Observational data: lunar laser ranging (LLR)
- > Change rate of gravitational constant G (by planetary dynamics)
 - Observational data: Earth-Mars, Earth-Mercury ranges to planet orbiters, other ranging data, optical data

Dynamical model of the Solar system includes:

- Point-masses interaction according to GR
- Sun's oblateness (J2) and Lense-Thirring acceleration
- Kinematic equations of the Earth's rotation and tides, acceleration and torque from non-spherical gravity up to degree 6
- Elastic Moon with tides and liquid core, non-spherical gravity up to degree 6

Observational data

Planets and natural satellites: optical, radar Spacecraft: ranging, VLBI, 3D Landers: randing (differential) Lunar retroreflector paners: laser ranging

Moon	31500 laser ranging 1970-2022	
Mercury	778 ranges 1964-1997	1161 + <mark>9</mark> MESSENGER 2011-2014
Venus	1249 ranges 1961-1996 18 Magellan 1990-1994	2 Cassini 1998-1999 1249+54 VEX 2006-2013
Mars		6143+15 MGS 1998-2006 3596 MEX 2009-2014
	1122+14952 Viking-1 90+7551 Pathfinder 1976-1982 1997	9748+111 Odyssey 2002-2017
	79 Viking-2 2 Phobos-2 1976-1977 1989	2477+<mark>75</mark> MRO 2006-2017
	12600+ optical (including 4 satellites) 1919-2018	
Jupiter	2 Pioneer 10, 11 2 Voyager 1, 2 1 Ulysses 1973, 1974 1979 1992	1 Cassini 4 Juno 2016-2017 2000
	24 Galileo 1996-1997	More than 200 000 in total
Saturn	14100+ optical (including 8 satellites) 1913-2015	
		147+15 Cassini 2004-2014
Uranus	19000+ optical (including 5 satellites) 1919-2016	
UTATIUS	I Voyager 2 1986	
Nontura	10500+ optical (including 2 satellites) 1899-2019	
Neptune	1 Voyager 2 1989	
Pluto	7300+ optical 1914-2013	

Planetary equations

- Einstein-Infeld-Hoffmann relativistic equations for point masses (Sun, Moon, planets, Pluto, 5 asteroids)
- Acceleration from solar oblateness
- Lense-Thirring acceleration

$$f_x = -\frac{3}{2}\mu J_2 \frac{R^2}{r^5} x \left(1 - 5\frac{z^2}{r^2}\right)$$
$$f_y = -\frac{3}{2}\mu J_2 \frac{R^2}{r^5} y \left(1 - 5\frac{z^2}{r^2}\right)$$
$$f_z = -\frac{3}{2}\mu J_2 \frac{R^2}{r^5} z \left(3 - 5\frac{z^2}{r^2}\right)$$

$$\begin{split} \vec{a}_{A} &= \sum_{B \neq A} \frac{Gm_{B} \vec{n}_{BA}}{r_{AB}^{2}} \\ &+ \frac{1}{c^{2}} \sum_{B \neq A} \frac{Gm_{B} \vec{n}_{BA}}{r_{AB}^{2}} \left[v_{A}^{2} + 2v_{B}^{2} - 4(\vec{v}_{A} \cdot \vec{v}_{B}) - \frac{3}{2} (\vec{n}_{AB} \cdot \vec{v}_{B})^{2} \\ &- 4 \sum_{C \neq A} \frac{Gm_{C}}{r_{AC}} - \sum_{C \neq B} \frac{Gm_{C}}{r_{BC}} + \frac{1}{2} ((\vec{x}_{B} - \vec{x}_{A}) \cdot \vec{a}_{B}) \right] \\ &+ \frac{1}{c^{2}} \sum_{B \neq A} \frac{Gm_{B}}{r_{AB}^{2}} \left[\vec{n}_{AB} \cdot (4\vec{v}_{A} - 3\vec{v}_{B}) \right] (\vec{v}_{A} - \vec{v}_{B}) \\ &+ \frac{7}{2c^{2}} \sum_{B \neq A} \frac{Gm_{B} \vec{a}_{B}}{r_{AB}} + O(c^{-4}) \end{split}$$

$$\ddot{\mathbf{r}}_i^{ ext{LT}} = rac{2}{c^2} GS_{ ext{Sun}} rac{1}{r_{iS}^3} R_{ ext{Sun}} \left(\dot{\mathbf{r}}_{iS} imes \mathbf{z} + 3 rac{\mathbf{z} \cdot \mathbf{r}_{iS}}{r_{iS}^2} \mathbf{r}_{iS} imes \dot{\mathbf{r}}_{iS}
ight)$$

Model of the Lunar Physical Libration

- Rigid body equations of rotation with an external torque
- Torque from point masses in lunar gravitational field: Earth, Sun, Venus, Mercury, Mars, Jupiter
- Figure-figure torque between Earth's J2 and Moon
- Dynamic tensor of inertia: delayed dissipation from rotation and Earth tides
- Inertia and torque from inner rotating liquid core
- Torque from friction on the core-mantle boundary

$$\mathbf{r}_{I} = \mathcal{R}_{z}(-\phi_{m})\mathcal{R}_{x}(-\theta_{m})\mathcal{R}_{z}(-\psi_{m})\mathbf{r}_{PA}$$

$$\dot{\phi}_{m} = (\omega_{m,x}\sin\psi_{m} + \omega_{m,y}\cos\psi_{m}) / \sin\theta_{m}$$
$$\dot{\theta}_{m} = \omega_{m,x}\cos\psi_{m} - \omega_{m,y}\sin\psi_{m}$$
$$\dot{\psi}_{m} = \omega_{m,z} - \dot{\phi}_{m}\cos\theta_{m}.$$

$$\dot{\boldsymbol{\omega}} = \mathbf{I}^{-1} \{ \Sigma_i \mathbf{N}_{fig-pm} + \mathbf{N}_{fig-fig} - \dot{\mathbf{I}} \boldsymbol{\omega} - \boldsymbol{\omega} \times \mathbf{I} \boldsymbol{\omega} \}$$

$$\mathbf{N}_{fig-fig} = \frac{15 \ \mu_e \mathcal{R}_e^2 J_{2e}}{2 \ r_e^5} \left\{ (1 - 7 \sin^2 \phi) [\mathbf{\hat{r}_e} \times \mathbf{I} \mathbf{\hat{r}_e}] \right. \\ \left. + 2 \sin \phi \ [\mathbf{\hat{r}_e} \times \mathbf{I} \mathbf{\hat{P}_e} + \mathbf{\hat{P}_e} \times \mathbf{I} \mathbf{\hat{r}_e}] \right. \\ \left. - \frac{2}{5} [\mathbf{\hat{P}_e} \times \mathbf{I} \mathbf{\hat{P}_e}] \right\}$$

$$\begin{split} \mathbf{I}_{m}(t) &= \tilde{\mathbf{I}}_{m} - \frac{k_{2,M} m_{E} R_{M}^{5}}{r^{5}} \begin{bmatrix} x^{2} - \frac{1}{3}r^{2} & xy & xz \\ xy & y^{2} - \frac{1}{3}r^{2} & yz \\ xz & yz & z^{2} - \frac{1}{3}r^{2} \end{bmatrix} \\ &+ \frac{k_{2,M} R_{M}^{5}}{3G} \begin{bmatrix} \omega_{m,x}^{2} - \frac{1}{3}(\omega_{m}^{2} - n^{2}) & \omega_{m,x}\omega_{m,y} & \omega_{m,x}\omega_{m,z} \\ \omega_{m,x}\omega_{m,y} & \omega_{m,y}^{2} - \frac{1}{3}(\omega_{m}^{2} - n^{2}) & \omega_{m,y}\omega_{m,z} \\ \omega_{m,x}\omega_{m,z} & \omega_{m,y}\omega_{m,z} & \omega_{m,z}^{2} - \frac{1}{3}(\omega_{m}^{2} + 2n^{2}) \end{bmatrix} \end{split}$$

$$\dot{\boldsymbol{\omega}}_{m} = \mathbf{I}_{m}^{-1} \left\{ \sum_{A \neq M} \mathbf{N}_{M, figM-pmA} + \mathbf{N}_{M, figM-figE} - \dot{\mathbf{I}}_{m} \boldsymbol{\omega}_{m} - \boldsymbol{\omega}_{m} \times \mathbf{I}_{m} \boldsymbol{\omega}_{m} + \mathbf{N}_{cmb} \right\} \quad \dot{\boldsymbol{\omega}}_{c} = \mathbf{I}_{c}^{-1} \left\{ -\boldsymbol{\omega}_{m} \times \mathbf{I}_{c} \boldsymbol{\omega}_{c} - \mathbf{N}_{cmb} \right\}$$

$$\mathbf{N}_{cmb} = k_{v} (\boldsymbol{\omega}_{c} - \boldsymbol{\omega}_{m}) + (C_{c} - A_{c}) (\hat{\mathbf{z}}_{m} \cdot \boldsymbol{\omega}_{c}) (\hat{\mathbf{z}}_{m} \times \boldsymbol{\omega}_{c})$$

$$\mathbf{I}_{c} = \alpha_{c} C_{T} \begin{bmatrix} 1 - f_{c} \\ 0 \\ 0 \end{bmatrix}$$



Determined parameters

Solar system parameters

- Orbital elements of planets at epoch
- ▷ Gm of the Sun, the Moon, and 277 asteroids
- Gms of the Main asteroid belt and Kuiper belt
- Orientation angles of ephemeris frame to ICRF
- 13 parameters of rotation of Mars
- Earth/Moon mass ratio
- Two angles of solar equator plane
- $\,\triangleright\,\,$ Solar oblateness factor J2 $_{\odot}$

Reduction parameters

- Corrections to reference surfaces of Venus, Mercury, and Mars
- 8 coefficients of topography of Mercury
- ▷ 5 phase corrections for outer planets
- Coordinates of 3 Martian landers
- Solar plasma delay factor
- Spacecraft ranging biases (MGS/Odyssey, MRO) for 15 DSN stations
- Corrections to spacecraft transponder delays
- Positions of lunar retroreflectors and lunar laser ranging reference points
- Corrections to LLR ranges (biases)

Lunar parameters

- Geocentric position and velocity of the Moon at epoch
- Lunar physical libration angles and their rates at epoch
- Ratios of the lunar moments of inertia
- Lunar core oblateness factor, core-mantle friction parameter
- Angular velocity of lunar core at epoch
- Lunar tidal delay
- \triangleright Love number h₂

Special parameters (determined in solutions together with the normal parameters)

- $\,\triangleright\,\,$ Change rate of ${\rm Gm}_{\odot}$
- ▷ m^g/mⁱ(Earth) m^g/mⁱ(Moon)

About 500 parameters in total

Equivalence principle

Universality of free fall:

$$\left(\frac{m^{\mathrm{g}}}{m^{\mathrm{i}}}\right)_{M} \stackrel{?}{=} \left(\frac{m^{\mathrm{g}}}{m^{\mathrm{i}}}\right)_{E}$$

Newtonian differential Earth-Moon acceleration in Sun's gravitational field:

$$\delta = \frac{m^{g}}{m^{i}} - 1 \qquad \mathbf{a}_{\mathrm{M}} - \mathbf{a}_{\mathrm{E}} = -\frac{G\left[(1 + \delta_{M})m_{M}^{g}\right] + (1 + \delta_{E})m_{E}^{g}\right]}{r_{\mathrm{EM}}^{3}}\mathbf{r}_{\mathrm{EM}} + Gm_{\mathrm{S}}^{g}\left[\frac{\mathbf{r}_{\mathrm{SE}}}{r_{\mathrm{SE}}^{3}} - \frac{\mathbf{r}_{\mathrm{SM}}}{r_{\mathrm{SM}}^{3}}\right] + Gm_{\mathrm{S}}^{g}\left[\frac{\mathbf{r}_{\mathrm{SE}}}{r_{\mathrm{SE}}^{3}}\delta_{E} - \frac{\mathbf{r}_{\mathrm{SM}}}{r_{\mathrm{SM}}^{3}}\delta_{M}\right] \mathbf{a}_{\mathrm{ME}}^{\mathrm{EP}} = Gm_{\mathrm{S}}^{g}\left[\frac{\mathbf{r}_{\mathrm{SE}}}{r_{\mathrm{SE}}^{3}}\delta_{E} - \frac{\mathbf{r}_{\mathrm{SM}}}{r_{\mathrm{SM}}^{3}}\delta_{M}\right] \approx Gm_{\mathrm{S}}^{g}\frac{\mathbf{r}_{\mathrm{SE}}}{r_{\mathrm{SE}}^{3}}\left[\delta_{E} - \delta_{M}\right]$$

Lunar solution estimate: $\delta_{E} - \delta_{M} = (10 \pm 30) \times 10^{-15}$ (3 σ error)

Conclusion: No sign of Earth's and Moon's accelerations in gravity of the Sun being different above the level of 3×10^{-14} (dimensionless units).

It is *probably* not possible to separate SEP and WEP using only the dynamics of the Solar system (see discussion in <u>Viswanathan et al. 2018</u>).

With WEP having been confirmed to better accuracy (<u>Touboul et al. 2022</u>), we can see the above as a bound for SEP.

Prior and determined asteroid masses



Modeling of delay in solar plasma



Spherically symmetric model

Ne ~ $1/r^2$ (Attempt to use the ENLIL solar wind data has not improved solution yet)

Medium-term variations of electron density in solar plasma

Electron density was calculated from OMNI data (<u>King &</u> <u>Papatashvili, 2005</u>) and then averaged to obtain yearly trends.

Single factor of solar plasma delay for 2009-2017

- The factor was regularized in the solution and came out equal to 1.034.
- > The solar plasma delay correlates with GM_{\odot} in the solution. The regularization allowed to improve the accuracy of GM_{\odot} by ~12%.



Effect of solar plasma delay



Estimates of $\frac{G}{c}$ using Solar System dynamics

(estimate GM_{\odot} and \dot{GM}_{\odot} from LSM) $GM_{\odot}(t) = GM_{\odot} + \dot{GM}_{\odot}t$ $\frac{\dot{GM}_{\odot}}{GM_{\odot}} = \frac{\dot{G}}{G} + \frac{\dot{M}_{\odot}}{M_{\odot}} = (-2.6 \pm 3.6) \times 10^{-14}$ $\frac{\dot{M}_{\odot}}{M_{\odot}} = \frac{1}{M_{\odot}} (\dot{M}_{\odot}^{\text{rad}} + \dot{M}_{\odot}^{\text{wind}} + \dot{M}_{\odot}^{\text{fall}})$

$$\frac{\dot{M}_{\odot}^{\text{rad}}}{M_{\odot}} = (-6.7 \pm 1.8) \times 10^{-14} \,\text{yr}^{-1} \,(3\sigma)$$
$$\frac{\dot{M}_{\odot}^{\text{wind}}}{M_{\odot}} = (-4.8 \pm 1.8) \times 10^{-14} \,\text{yr}^{-1} \,(3\sigma)$$
$$\frac{\dot{M}_{\odot}^{\text{fall}}}{\dot{M}_{\odot}} = (0.5 \pm 0.5) \times 10^{-14} \,\text{yr}^{-1} \,(3\sigma)$$

Mass lost as a result of thermonuclear fusion. (IAU Resolution B3, 2015).

Mass carried away with the solar wind. Data from the Ulysses spacecraft (NASA). Calculations from (<u>Pitjeva et. al. 2021</u>)

Asteroids falling onto Sun. Calculations from (<u>Pitjeva et. al. 2021</u>)

$$\frac{\dot{G}}{G} = (0.84 \pm 0.44) \times 10^{-13} / \text{yr}$$
 (3 σ)

Secular change of G: discussion

Other modern estimates made outside the Solar system:

- (-6 ± 11) × 10⁻¹³/yr (<u>Zhu et al, 2015</u>) [pulsar period]
- > $(-36..45) \times 10^{-13}$ /yr (<u>Alvey et al, 2020</u>) [baryon density from CMB data]
- ➤ (0.53 ± 0.60) × 10⁻¹³/yr (<u>Le, 2021</u>) [quasar spectra]
- ➤ (0.0092 ± 0.028) × 10⁻¹³/yr (<u>Le, 2024</u>) [quasar spectra]

Estimate by Solar system dynamics

- > $(-0.29 .. 0.46) \times 10^{-13}$ /yr (<u>Pitieva et al, 2021</u>) [preceding work by same group]
- > (0.84 ± 0.44) × 10⁻¹³/yr (this work)

Low-frequency gravitational waves (<u>Agazie et al. 2023</u>) [pulsar timing arrays]. Nano-Hz frequencies (= periods of decades).